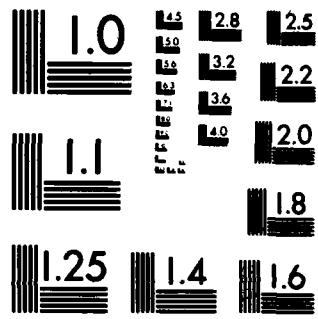


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RADC-TR-83-56  
Final Technical Report  
March 1983

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## **MODIFICATIONS TO ACROSS MODEL #2 DESIGN**

**The Charles Stark Draper Laboratory, Inc.**

**Sponsored by  
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MODIFICATIONS TO ACROSS MODEL #2 DESIGN

Timothy Henderson

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specifications will be designed for this model. This design will be compared with that based on the second VCOSS model. This model is a maximum mass, stiffness controlled design. The purpose of this design is to reduce control system requirements by stiffening the structure in order to raise critical modal frequencies. The MSC/NASTRAM finite element models for all three designs are listed in the Appendices.

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## SECTION 1

### INTRODUCTION

In recent years the development of the technology for active control of structural vibrations of Large Space Structures (LSS) has been addressed in the Active Control of Space Structures (ACOSS) program. Early in the program a need for a common design model to allow evaluation and comparison of the various approaches was identified. The first design model, ACOSS Model 1, was developed in 1979. This structure, a simple ground attached tetrahedron, provided a means of evaluating control designs but lacked any real physical relationship with typical LSS. Because of the desire for an evaluation model which had dimensions, materials, optical system, and performance more closely related to actual structures, ACOSS Model #2 was developed in May 1980. In order to meet the design goal of a low-order structural model, many simplifying assumptions were made in the modelling of the optical system mirrors and the supporting structure. This was consistent with the desire to use the model to evaluate control design methods without concern for detailed mechanical design of the components. This model has provided a simple means of comparing and evaluating the control designs of the various ACOSS contractors. A complete description of the model along with a set of disturbances is contained in Reference 1.

The Vibration Control of Space Structures (VCOSS) program is intended to study the application of ACOSS technology to an actual spacecraft design. This includes specification of sensor and actuator hardware and assessment of the performance of the system including these components. The system model will be based on ACOSS Model #2, but with modifications to reduce the structural mass to the minimum required to maintain structural stability. This system will rely entirely upon the control system for vibration suppression to meet the performance requirements. This will be compared with another model which is also a modification of ACOSS Model #2. This second design will be a strengthened version of Model #2. The goal of this design is to meet the performance specifications by passive means: structural stiffening and passive damping. The structure will be stiffened by increasing the sizes of the existing members in Model #2 until the total system mass is equal to the maximum cargo capacity of the shuttle into polar orbit, 15000 kg. No additional structural members will be added so that the only difference between the two designs will be the size of the members.

The first step in modifying ACOSS Model #2 to generate the two models required for the VCOSS program was to reassess the basic assumptions which were used in the design. This is necessary because in order to establish the hardware requirements of a vibration control system the system model must accurately reflect the mechanical properties of an actual system. In this structure the important properties are the interaction of the rigid mirrors with the flexible support and the interface

between the equipment section and the optical support truss. In the original design of ACOSS Model #2, the goal of minimizing the total number of degrees-of-freedom in the system resulted in a simple method of including the mirror mass effects. This approach did not include details of the kinematic connection of the mirrors to the structure or of the motion of the mirror centroids. The modelling of the equipment section did not provide adequate space for the isolation system which will separate the two bodies.

Section 2 describes an updated version of ACOSS Model #2. This update reflects the need for a more detailed model which more closely reflects the behavior of an actual spacecraft and can be used as a basis for the hardware implementation studies required in the VCOSS program. In this updated design the original geometry and stiffness are retained as much as possible. Changes were made to the structural model to include the rigid body inertia properties of each mirror and a detailed kinematic mount connecting it to the support structure. In addition, the model of the equipment section was changed to include a more representative mass distribution and to provide clearance for the isolation system hardware.

Using this updated design as a baseline, the two VCOSS models were generated. Section 3 describes the minimum mass, strength-controlled design. In this model, the size of each structural member has been reduced to the minimum allowed by constraints on local buckling, member natural frequency and minimum wall thickness for the tube sections. The updated finite element model and natural frequency tables are listed.

The stiffness-controlled design is described in Section 4. Each structural member has been increased in order to uniformly increase the stiffness of the structure. A limit of 15000 kg has been placed on the total mass of the system to allow for placement into a polar orbit by the shuttle. The updated finite element model and natural frequency tables are given.

In order to avoid confusion between the original ACOSS Model #2 and the three new versions of it which are described in this report, a system of revision numbers has been established. This will provide a means of easily identifying the existing versions as well as all those which may be generated in the future.

Revision 0: The original ACOSS Model #2, first presented in May 1980.

Revision 1: Updated version of the original design which includes more detailed mirror and equipment section modes. This design is described in Section 2.

Revision 3: The strength controlled, minimum mass design based on Revision 1. This is the VCOSS actively controlled design. Details of the model are presented in Section 3.

Revision 4: The stiffness controlled maximum mass design based on Revision 1. This design is presented in Section 4.

Reference 1

RADC-TR-80-377, Interim Report, Jan 1981, "ACOSS Six (Active Control of Space Structures)

## SECTION 2

### UPDATED STRUCTURAL DESIGN

#### 2.1 Introduction

The baseline design for the two VCOSS models is the original ACOSS Model #2 shown in Figure 2.1. This model was generated in May 1980 to provide a simple, unclassified tool to evaluate the active control philosophies proposed by a variety of sources. As the needs of the users of the model have changed, the model has been changed to provide the required fidelity. An updated ACOSS Model #2 is presented in this section. Stiffness controlled and strength controlled versions of this updated model will be described in following chapters. The design changes which were incorporated into this update reflect the need to provide better correlation between this simplified model and typical large space structures.

The changes which were made to Model #2 are confined to two areas, the details of the mirror support structures and the equipment section. The metering truss, which separates the upper and lower mirrors, the solar panels, and the isolation springs are unchanged from the original design. The modifications to the structural design and the finite element model are described in the following sections. The NASTRAN input data for this model is listed in Appendix A. The finite element model for Revision 1 is shown in Figure 2.2.

In order to facilitate these modifications and the structural modifications in the VCOSS designs, the structural and non-structural mass have been uncoupled. In Revision 0, the structural mass was added to the non-structural mass and lumped at the mirror support points. In the new model, the non-structural mass is lumped at the mirror centroids and support points and the structural mass is lumped at all node points. The structural mass will be computed automatically by NASTRAN using the length and area of each member and the material density of  $1720 \text{ kg/m}^3$ . The structural and non-structural mass at each node point and the system mass properties for Revision 1 are listed in Table 2.1.

#### 2.2 Mirror Design Modifications

Modifications were made to the models of the mirrors in the system in order to reflect, in detail, the interaction between the rigid mirrors and the flexible optical support truss. In the original design, (Revision 0), in which the number of lumped masses was kept to a minimum, it was assumed that the masses of each mirror and its supporting structure were evenly distributed to the support points without regard to the detail of the connection.

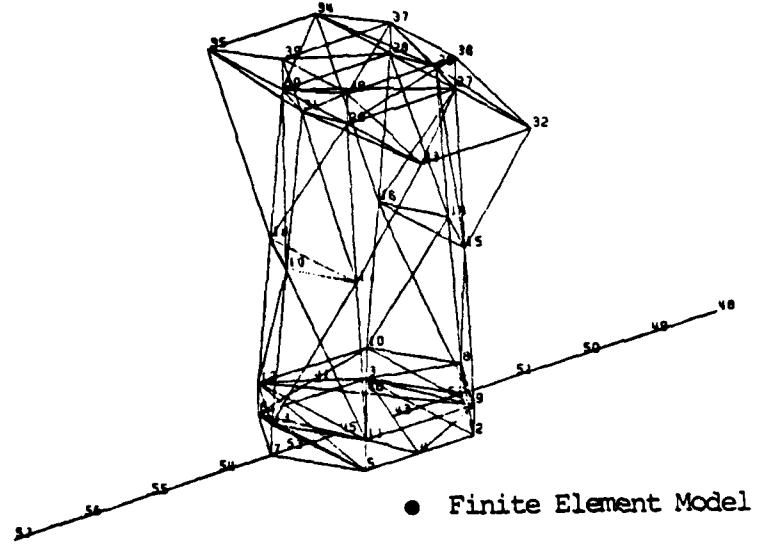
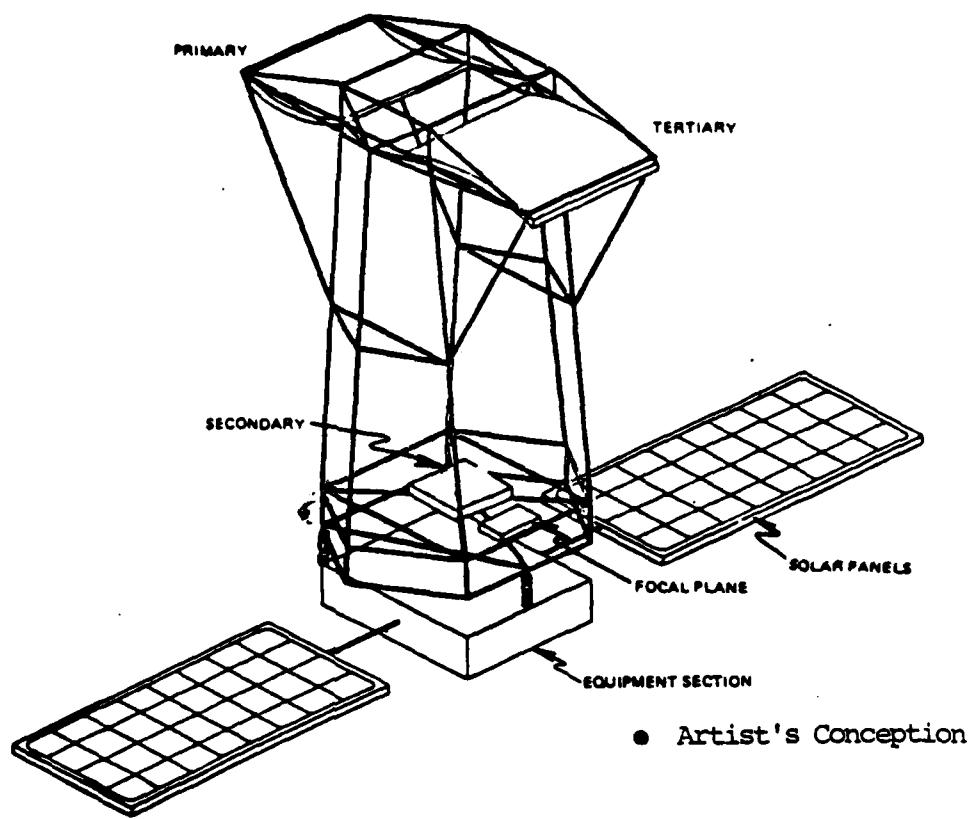


Figure 2.1. ACOSS Model #2 (Revision 0)

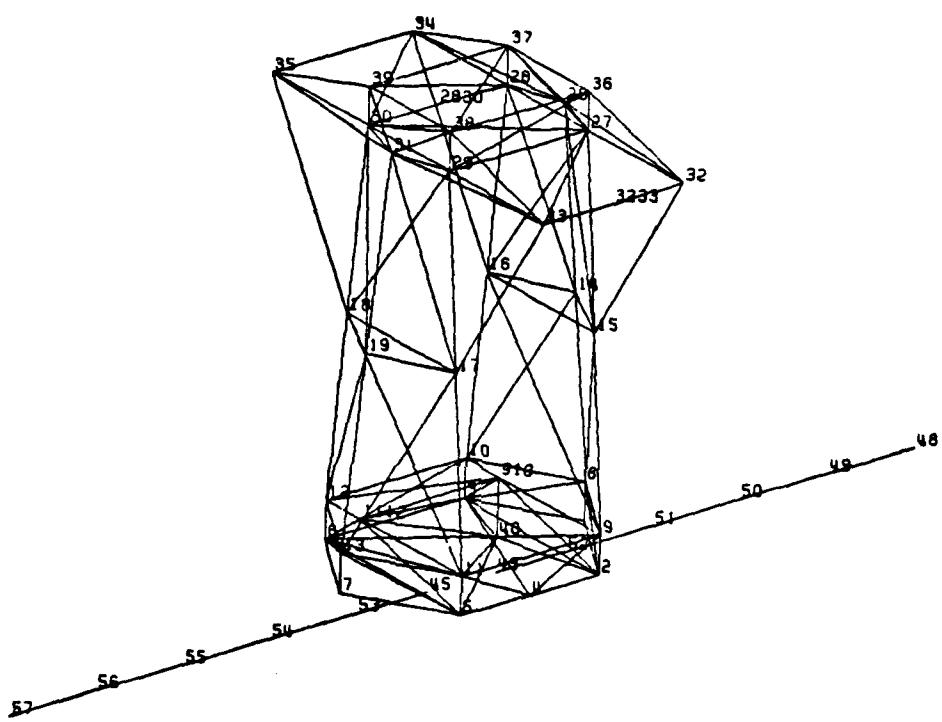


Figure 2.2: ACOSS Model #2 (Revision 1)

Table 2.1. Lumped mass distribution

node	structural	non-struct.	node	structural	non-struct.
	mass (kg)	mass (kg)		mass (kg)	mass (kg)
1	7.34		36	9.99	
2	21.97		37	12.17	
3	34.03		38	12.17	
4	20.86		39	9.99	
5	21.97		40	26.85	
6	33.83		42	0.0	
7	7.34		43	3.24	
8	21.13		44	0.0	3500.
9	25.40	67.4	45	3.24	
10	21.50	67.4	46	0.0	
11	25.40	67.4	47	0.0	
12	21.57	67.4	48	4.05	81.91
13	21.13		49	8.09	
14	16.59		50	8.09	163.82
15	21.48		51	8.09	
16	22.71		52	7.28	73.82
17	21.48		53	7.28	73.82
18	22.71		54	8.09	
19	16.59		55	8.09	163.82
26	17.53		56	8.09	
27	23.00	69.50	57	4.05	81.91
28	45.94	6.74	910	23.51	
29	17.25	69.50	1001		1000.0
30	51.70	6.74	1002		800.0
31	17.53		1003		1200.0
32	43.51	6.74	1004		600.0
33	47.09	6.74	1112	23.64	
34	18.41	69.50	2330	62.76	
35	14.83	69.50	3233	62.76	
<b>Totals</b>			1023.34	3313.66	

Total mass = 9337 kg

Center of Mass Location  
x = 0.0 m

y = -0.237 m

z = 6.983 m

The new mirror models assume that each surface is a planar rigid body which is connected to the support structure by kinematic mounts. This type of mount can only transmit rigid body motion between the support structure and the mirror. Elastic motion of the support points relative to each other will not cause any deformation of the mirror surface. The finite element model has been changed so that the translational and rotational inertia properties of each mirror are lumped at a node point at its center of mass. This node is connected to the three support points of the surface by rigid elements which are attached at the six degrees of freedom required for the kinematic mount. The details of a kinematic mount are shown in Figure 2.3. In the drawing, point A is supported in the x, y, and z directions, point B is supported in the y and z directions, and point C is supported in the z direction.

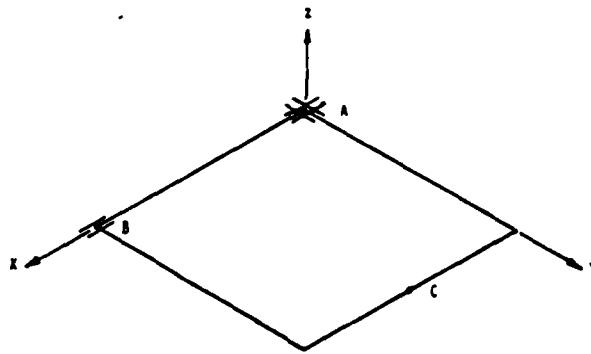


Figure 2.3: Typical kinematic mount

The new model of each mirror will be described in the following sections. A summary of these models is given in Table 2.2.

#### 2.2.1 Primary Mirror

The primary mirror is one of the two large mirrors at the top level of the optical support truss. In the original design (Revision 0) masses were lumped at nodes 34, 35, 28, and 30 to represent the structural and non-structural mass associated with this surface. The motion of the mirror was defined by a kinematic mount with point A at 34, point B at node 35, and point C located midway between nodes 28 and 30. The displacement at point C was interpolated from the two nodal values. This kinematic mount was not included in the finite element model but was used only in the line-of-sight error model.

In the new design, two new node points are added to the finite element model. One, node 1001, is located at the center of mass. The second, node 2830, lies midway between nodes 28 and 30 to provide the third

support point for the kinematic mount. The beam element connecting nodes 28 and 30 and containing node 2830 has been strengthened to support the mass of the mirror. The stiffened beam is a tubular truss as shown in Figure 2.4. It is designed to prevent significant line-of-sight errors due to bending at frequencies below 40 Hz. The center of mass is connected to the support points by rigid elements which are constrained in the six degrees of freedom required for the kinematic mount. The new model of the primary mirror is shown in Figure 2.5.

#### 2.2.2 Secondary Mirror

The motion of the secondary mirror was defined in Revision 0 by the six degrees of freedom (three translations and three rotations) at node 40. The mass was lumped at the edge nodes of the lower support truss so that the motion of the mirror plane was equal to the motion of node 40. In the new model, the mass of the secondary will be lumped at its center of mass, node 1002, which will be attached to the supporting structure by a kinematic mount as shown in Figure 2.6. The structural design was altered to include the additional support points for the secondary. The finite element model was changed by adding the two support point nodes, node 910 and node 1112, moving node 40, and including the members required to brace the supports. Point A of the mount will be node 910, point B will be node 1112, and point C will be node 40. Rigid bars provide the connection between the mount degrees of freedom and the center of mass.

#### 2.2.3 Tertiary Mirror

The changes in the model of the tertiary mirror are similar to those made in the model of the primary. The center of mass is located at node 1003. It is kinematically mounted on the support truss by connections to node 27 (point A), node 29 (point B), and node 3233 (point C). Node 3233 is located midway between nodes 32 and 33 on the stiffened beam connecting them. The new configuration of the tertiary mirror model is shown in Figure 2.5.

#### 2.2.4 Focal Plane

The only change made to the model of the focal plane consists of the addition of the kinematic mount. The mass properties of the focal plane are lumped at its center of mass, node 1004. Rigid bars connect it to the three support points, node 11 (point A), node 9 (point B), and node 40 (point C). The new configuration is shown in Figure 2.6.

Table 2.2. Modified Mirror Models

Primary:

Mass = 1000 kg  
 $I_x = 4083.33 \text{ kg-m}^2$   
 $I_y = 5333.33 \text{ kg-m}^2$   
 $I_z = 9416.67 \text{ kg-m}^2$

Center of Mass: node 1001

Support Points:

Node 34 dof 1,2,3  
 Node 35 dof 2,3  
 Node 2830 dof 3

Secondary:

Mass = 800 kg  
 $I_x = 1666.67 \text{ kg-m}^2$   
 $I_y = 4266.67 \text{ kg-m}^2$   
 $I_z = 5933.33 \text{ kg-m}^2$

Center of Mass: node 1002

Support Points:  
 Node 910 dof 1,2,3  
 Node 1112 dof 2,3  
 Node 40 dof 3

Tertiary:

Mass = 1200 kg  
 $I_x = 4900.00 \text{ kg-m}^2$   
 $I_y = 6400.00 \text{ kg-m}^2$   
 $I_z = 11300.0 \text{ kg-m}^2$

Center of Mass: node 1003

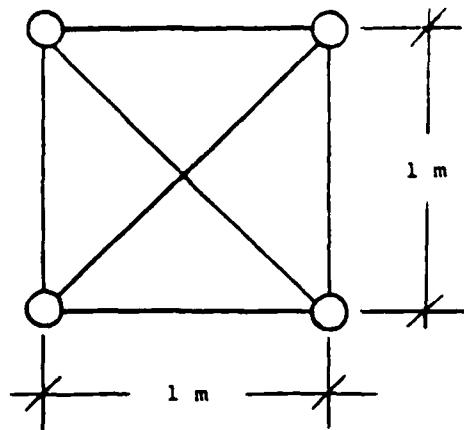
Support Points:  
 Node 27 dof 1,2,3  
 Node 29 dof 2,3  
 Node 3233 dof 3

Focal Plane:

Mass = 600 kg  
 $I_x = 200.00 \text{ kg-m}^2$   
 $I_y = 800.00 \text{ kg-m}^2$   
 $I_z = 1000.00 \text{ kg-m}^2$

Center of Mass: node 1004

Support Points:  
 Node 11 dof 1,2,3  
 Node 9 dof 2,3  
 Node 40 dof 3



Tube Properties:

$$\text{Radius} = 0.11 \text{ m}$$

$$t = 0.0022 \text{ m}$$

$$\text{Area} = 0.00152 \text{ m}^2$$

Section Properties:

$$\text{Area} = 4 A_{\text{tube}} = 0.00608 \text{ m}^2$$

$$I = 0.00152 \text{ m}^4$$

$$J = 0.00304 \text{ m}^4$$

$$\text{Mass/L} = 15.69 \text{ kg/m}$$

Figure 2.4. Stiffened mirror support beam

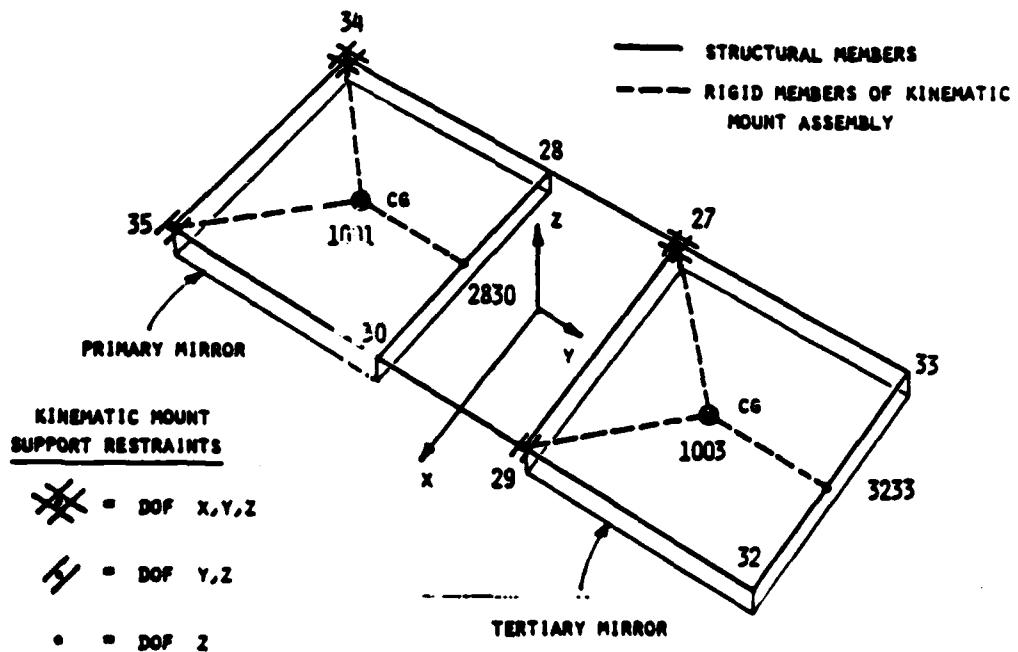


Figure 2.5. Modified primary and tertiary mirror models

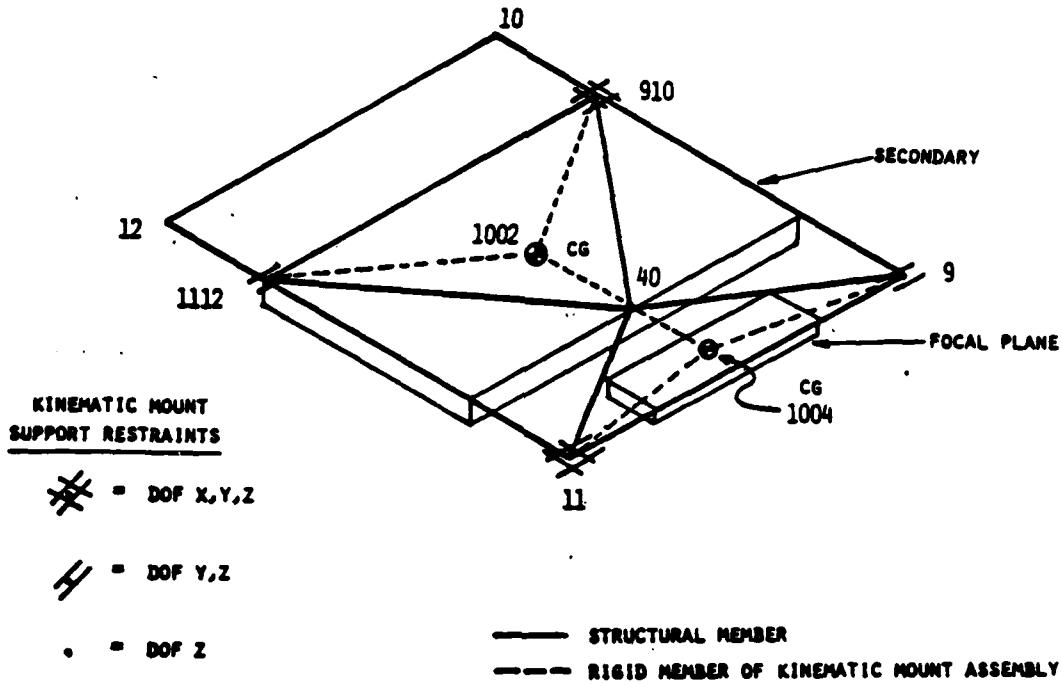


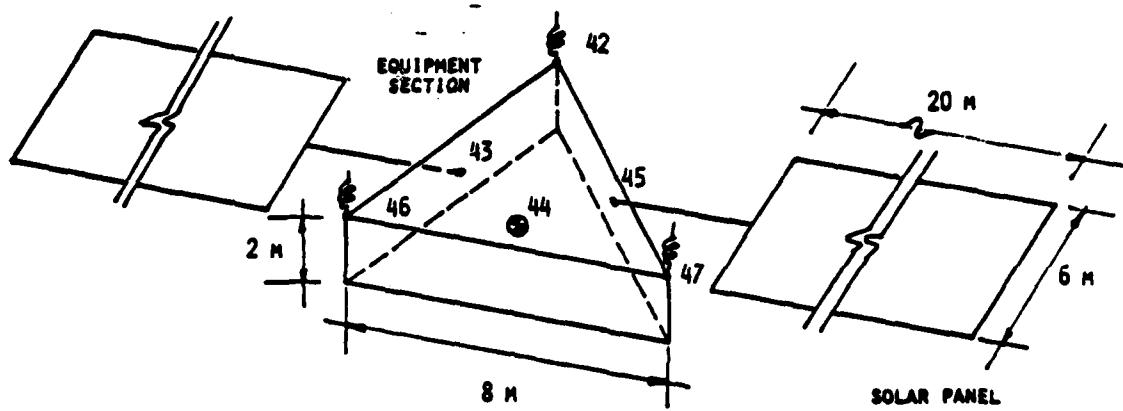
Figure 2.6. Modified secondary and focal plane mirror models

### 2.3 EQUIPMENT SECTION MODIFICATIONS

Changes were also made to the model of the equipment section. In the original design the equipment section was modelled as a rigid plate and located at the same point on the z-axis as the lower chord of the optical support truss. The two sections were connected by springs as part of the isolation system. While this configuration is mathematically correct, in an actual system the springs will have a finite length. The equipment section will now be modelled as a uniform rigid body, triangular in shape and two meters in depth. The equipment section and the optical support truss will be separated by thirty centimeters to allow room for the isolation system. The new mass properties and location of the equipment section are given in Figure 2.7.

### 2.4 LINE-OF-SIGHT-ERROR MODEL CHANGES

Due to changes in the models of the mirrors and their support points, a few minor changes are necessary in the implementation of the line-of-sight error model for Revision 1 and all later versions. This



- EQUIPMENT SECTION MODELLED AS RIGID BODY

Mass = 3500 kg

Center of Mass (node 44)

$$I_{xx}^2 = 20611.11 \text{ kg-m}^2$$

$$x = 0.0 \text{ m}$$

$$I_{yy}^2 = 10500.00 \text{ kg-m}^2$$

$$y = -1.67 \text{ m}$$

$$I_{zz}^2 = 28777.77 \text{ kg-m}^2$$

$$z = -1.30 \text{ m}$$

Figure 2.7. Equipment section properties

error model relates the rotation about the x and y axes and the defocus of the line-of-sight to the displacements of the mirror support point nodes in the finite element model. Full details about the theory and implementation of the error model are given in CSDL Report C-5437. The changes in the updated version are due to the addition of the nodes 2830 and 3233 in the kinematic mount models for the primary and tertiary mirrors and the addition of the kinematic mount for the support of the secondary. The updated equations relating LOS errors to node point displacements are:

$$\begin{aligned} \text{LOSX} = & - (.0186)Y_{34} - (.1429)Z_{34} - (.0186)Y_{35} - (.1429)Z_{35} \\ & + (.2857)Z_{2830} + (.0807)Y_{27} - (.3549)Z_{27} + (.0807)Y_{29} \\ & - (.3549)Z_{29} + (.7098)Z_{3233} - (3.4842)\theta X_{1002} - (.0621)Y_{11} \\ & - (.0621)Y_9 \end{aligned}$$

$$\begin{aligned} \text{LOSY} = & - (.0371)X_{34} - (.0464)Y_{34} - (.2500)Z_{34} + (.0464)Y_{34} \\ & + (.2500)Z_{35} + (.1613)X_{27} - (.0605)Y_{27} - (.6211)Z_{27} \\ & + (.0605)Y_{29} + (.6211)Z_{29} + (3.4842)\theta Y_{1002} - (.1242)X_{11} \\ & - (.0776)Y_{11} + (.0776)Y_9 \end{aligned}$$

$$\begin{aligned} \text{DEFOCUS} = & - (.0191)Z_{34} - (.0191)Z_{35} + (.1274)Z_{2830} \\ & + (.7780)Z_{27} + (.7780)Z_{29} - (.4668)Z_{3233} - (2.000)Z_{40} \\ & - (.1785)Z_{1002} + (.5000)Z_9 + (.5000)Z_{11} \end{aligned}$$

Since this model is a linear function of the nodal displacements it is included as a set of constraint equations in the NASTRAN finite element model. The three line-of-sight components are assigned to an added node which is not part of the structural model. In all revisions of ACOSS Model #2 the line-of-sight errors are included at node 100 so that:

```
node 100 dof 1 = losx  
node 100 dof 2 = losy  
node 100 dof 3 = defocus
```

## 2.5 ANALYSIS

A normal modes analysis of the finite element model containing all of the changes, Revision 1, was performed using the MSC/NASTRAN finite element program. The resulting frequencies are shown in Table 2.3. Also listed in the table are the components of the line-of-sight errors for each mode, node 100. These are based on modes which have been normalized to a unit generalized mass.



## SECTION 3

### STRENGTH CONTROLLED DESIGN

#### 3.1 Introduction

The strength controlled design (Revision 3) was created by resizing the structural members in the updated ACOSS Model #2 (Revision 1) to meet the minimum requirements for local stability and frequency. The non-structural mass due to the mirrors and solar panels, which represents 90% of the total system mass in revision 1, is unchanged in this design. Because of this, even though the structural mass was reduced by 50% the total mass was reduced by only 373 kg to 8963 kg. This results in a very flexible structure with very low natural frequencies.

The design constraints were established to prevent local failure of individual elements in the structure due to buckling or excessive dynamic interaction due to low member natural frequency. The local buckling load,  $P_{cr}$ , is directly related to the slenderness ( $l/r$ ) of each beam by the formula:

$$P_{cr} = \frac{\pi^2 EA}{(K + \frac{1}{r})^2}$$

since the element forces are expected to be small, a slenderness limit of  $l/r = 400$  was used, and K equals .7 to account for end fixity. Using this equation, the minimum buckling load in the structure is 1060N. The natural frequency of each member was constrained to be greater than 10 Hz to prevent significant interaction between system vibrations and local vibrations. Finally, the wall thickness of the tubes was constrained to be a minimum of .03 cm. The NASTRAN input deck is listed in Appendix B. The tubular truss elements used to support the primary and tertiary mirrors were not changed. Reducing the size of these elements would have resulted in an unsound structure design due to the very low frequency local vibrations of these mirrors. These local effects would not be present in actual systems and would yield misleading results for this test model. The mass properties of this design are listed in Table 3.1.

### 3.2 Analysis

A normal modes analysis of the finite element model containing all of the changes, Revision 3, was performed using the MSC/NASTRAN finite element program. The resulting frequencies are shown in Table 3.2 along with the LOS errors for each mode. These LOS errors are based on a unit amplitude for each mode assuming they have been normalized to a unit generalized mass.

Table 3.1. Lumped mass distribution

node	structural	non-struct.	node	structural	non-struct.
	mass (kg)	mass (kg)		mass (kg)	mass (kg)
1	.80		36	2.68	
2	5.37		37	4.41	
3	10.72		38	4.41	
4	7.35		39	2.68	
5	5.37		40	9.48	
6	10.67		42	0.0	
7	.80		43	2.10	
8	7.59		44	0.0	3500.
9	14.12	67.4	45	2.10	
10	8.70	67.4	46	0.0	
11	14.12	67.4	47	0.0	
12	8.72	67.4	48	2.62	81.91
13	7.59		49	5.25	
14	12.12		50	5.25	163.82
15	17.25		51	5.25	
16	25.78		52	4.73	73.82
17	17.25		53	4.73	73.82
18	25.78		54	5.25	
19	12.12		55	5.25	163.82
26	10.78		56	5.25	
27	15.50	69.50	57	2.62	81.91
28	38.49	6.74	910	3.57	
29	10.80	69.50	1001		1000.0
30	43.19	6.74	1002		800.0
31	10.78		1003		1200.0
32	39.20	6.74	1004		600.0
33	42.17	6.74	1112	3.60	
34	11.51	69.50	2830	62.76	
35	8.54	69.50	3233	62.76	
Totals				649.97	8313.66

Total mass = 8963 kg

Center of Mass Location

x = 0.0 m

y = -0.240 m

z = 6.991 m



## SECTION 4

### STIFFNESS CONTROLLED DESIGN

#### 4.1 Introduction

The stiffness controlled structural design (Revision 4) is one component of the passively controlled system. In theory, the penalty paid for the increased mass of the stiffened system will be offset by a lighter, less complex passive control system. This will be true if the effect of the disturbances is reduced significantly by local stiffening or increasing the natural frequencies of the system.

The design of a stiffened structure can be approached in a number of ways depending on the desired results. The simplest approach is to use a trial and error method based on engineering judgement to increase member sizes and natural frequencies. At the other end of the spectrum are optimization techniques in which a performance index, which is a function of structural parameters, is optimized subject to constraints on the variables. The computation requirements for this type of approach can be extensive. The approach which was used to generate this model falls between these two extremes.

The objective of the redesign of this system is to improve the performance by reducing the effects of the disturbances. This is complicated by the fact that most of the elastic modes, except for those of the solar panels, will be excited by the disturbances and cause errors. Normal improvement approaches employing frequency separation between the disturbances and the structural modes will not be effective because of the high bandwidth of these forces. It will be impossible to add sufficient stiffness to raise the low frequency bending modes above 15 Hz. The maximum achievable frequency of these modes may be well below this point. Based on the considerations, the goal of this redesign will be to raise all frequencies of the optical support truss as much as possible and to raise the solar panel modes out of the bandwidth of the isolator modes.

Starting with the baseline design (Revision 1) a maximum mass model was generated by increasing the size of all members to the maximum permitted by the constraints.

$$\text{Radius/thickness} = 50$$

$$\text{Length/radius} = 40$$

$$\text{Maximum thickness} = 1.0 \text{ cm}$$

The maximum allowable area based on these constraints is the minimum of:

$$A_{MAX1} = 2\pi \cdot L/40 \cdot L/40 \cdot 1/50 = 7.854 \times 10^{-5} L^2$$

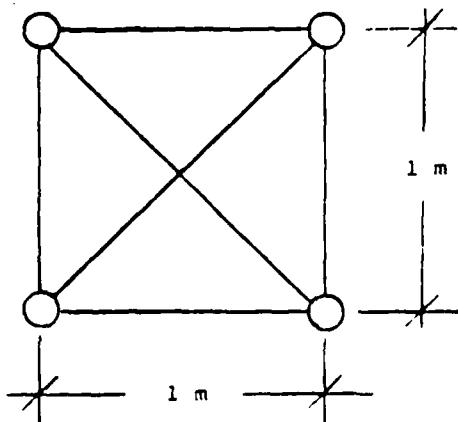
$$A_{MAX2} = 2\pi \cdot L/40 \cdot .01 = 1.5708 \times 10^{-3} L$$

$A_{MAX1}$  controls for all members in this model since it has the minimum value for elements with length less than twenty meters. The structural mass required for the maximum mass design is

$$\text{Mass} = \sum A_i L_i \rho$$

This must be scaled to meet the structural mass constraint of 6686 kg and total constraint of 15000 kg. All areas will be scaled uniformly to meet these constraints. Since the member areas are proportional to  $L^2$ , the stiffness matrix in both the axial and bending directions is proportional to the length  $L$ . If there is a wide range of lengths in the model, the shorter elements may not have sufficient strength. An analysis of this design showed significant strain energy in the short elements of the lower optical support truss. This was corrected by modifying the mass scale factor to account for the inverse of the length. This will result in areas which are proportional to the length and axial stiffness which is independent of length and equal for all elements. Analysis results using these modified section properties showed a significant improvement in frequency and a better distribution of strain energy.

To raise the frequencies of the solar panels above 1 Hz, and avoid overly large sections, the solar panel support booms were redesigned as a truss. The design, shown in Figure 4.1, is modelled by an equivalent beam in the finite element model.



Tube Properties:

$$\text{Radius} = 0.05 \text{ m}$$

$$t = 0.001 \text{ m}$$

$$\text{Area} = 0.000314 \text{ m}^2$$

Section Properties:

$$\text{Area} = 4 A_{tube} = 0.001257 \text{ m}^2$$

$$I = 0.000314 \text{ m}^4$$

$$J = 0.000628 \text{ m}^4$$

$$\text{Mass/L} = 3.82 \text{ kg/m}$$

Figure 4.1. Solar panel support boom design

The section properties of the members in this design are listed in Appendix C. The structural and non-structural mass lumped at each node point are listed in Table 4.1.

#### 4.2 Analysis

A normal modes analysis of the finite element model of the stiffness controlled design, Revision 4, was performed using the MSC/NASTRAN finite element program. The resulting frequencies are shown in Table 4.2 along with the LOS error for each mode. These LOS errors are based on a unit amplitude for each mode assuming they have been normalized to a unit generalized mass.

Table 4.1. Lumped mass distribution - Revision 4

node	structural	non-struct.	node	structural	non-struct.
	mass (kg)	mass (kg)		mass (kg)	mass (kg)
1	29.22		36	96.98	
2	113.01		37	126.20	
3	209.68		38	126.20	
4	123.35		39	96.98	
5	113.01		40	134.51	
6	207.15		42	0.0	
7	29.22		43	11.96	
8	151.76		44	0.0	3500.
9	195.78	67.4	45	11.96	
10	164.84	67.4	46	0.0	
11	195.78	67.4	47	0.0	
12	165.76	67.4	48	14.96	81.91
13	151.76		49	29.91	
14	196.39		50	29.91	163.82
15	269.43		51	29.91	
16	305.14		52	26.92	73.82
17	269.43		53	26.92	73.82
18	305.14		54	29.91	
19	196.39		55	29.91	163.82
26	189.90		56	29.91	
27	270.24	69.50	57	14.96	81.91
28	185.98	6.74	910	125.08	
29	195.18	69.50	1001		1000.0
30	261.05	6.74	1002		800.0
31	189.90		1003		1200.0
32	168.94	6.74	1004		600.0
33	214.79	6.74	1112	126.70	
34	209.38	69.50	2830	62.76	
35	163.52	69.50	3233	62.76	
<b>Totals</b>			<b>6686.40</b>	<b>8313.66</b>	

Total mass = 15000 kg

#### Center of Mass Location

$$x = 0.0 \text{ m}$$

$$y = -0.201 \text{ m}$$

$$z = 3.833 \text{ m}$$



## SECTION 5

### SUMMARY

In order to meet the needs of the VCOSS program for two models which represent the extremes of structural stiffness and were detailed enough to allow specification of control system hardware, modifications were made to the design of ACOSS Model #2. Three new revisions of this model were created and are described in this report. The MSC/NASTRAN input for each revision is included in an Appendix.

The first model, Revision 1, is an updated version of the original design, Revision 0. The design has been changed to add more detail to the models of the mirrors, mirror support structure and equipment section. The basic geometry and the stiffness of the structural members were not changed from the original ACOSS Model #2. The dynamic characteristics of this model are close to those exhibited by Revision 0.

Using this updated model as a baseline, two new revised models were created. These two new revised models, a lightweight, flexible design (Revision 3) and a heavy, stiff design (Revision 4) can be used to study the trade-offs between structural mass and control system complexity.

**APPENDIX A**

ID DRAPER,MODEL2  
 SOL 3  
 CHKPNT YES  
 TIME 10  
 CENO  
 TITLE = \*\*\*\*\* ACOS MODEL #2 \*\*\*\*\*  
 SUBTITLE = UPDATED ORIGINAL MODEL - REV 1  
 MPC = 100  
 METHOD = 600  
 DISP = ALL  
 \$ESE =ALL  
 BEGIN BULK  
 PARAM,USETPRT,1  
 PARAM GRDPNT 0 200 +10  
 EIGR 600 GIV  
 +10 MASS  
 \$  
 \$ KINEMATIC MOUNT: TERTIARY MIRROR  
 \$  
 RBE1,1003,27,123,29,23,3233,3,,+RB31  
 +RB31,UM,1003,123456  
 \$  
 \$ KINEMATIC MOUNT: PRIMARY MIRROR  
 \$  
 RBE1,1001,34,123,35,23,2830,3,,+RB11  
 +RB11,UM,1001,123456  
 \$  
 \$ KINEMATIC MOUNT: FOCAL PLANE  
 \$  
 RBE1,1004,11,123,9,23,40,3,,+RB41  
 +RB41,UM,1004,123456  
 \$  
 \$ KINEMATIC MOUNT: SECONDARY MIRROR  
 \$  
 RBE1,1002,910,123,1112,23,40,3,,+RB21  
 +RB21,UM,1002,123456  
 \$  
 \$ RIGID EQUIPMENT SECTION  
 \$  
 RBE2 141 44 123456 42 43 45 46 47  
 \$  
 \$ NODE POINT LOCATIONS  
 \$ NODE # X(M) Y(M) Z(M)  
 \$  
 GRID 1 -7.0 0.0 0.0  
 GRID 2 -4.0 5.0 0.0  
 GRID 3 -4.0 -5.0 0.0  
 GRID 4 0.0 5.0 0.0  
 GRID 5 4.0 5.0 0.0  
 GRID 6 4.0 -5.0 0.0  
 GRID 7 7.0 0.0 0.0  
 GRID 8 -7.0 0.0 2.0  
 GRID 9 -4.0 5.0 2.0  
 GRID 1004 0.0 4.0 2.0  
 GRID 10 -4.0 -5.0 2.0  
 GRID 910 -4.0 -2.5 2.0  
 GRID 1112 4.0 -2.5 2.0

GRID	11		4.0	5.0	2.0
GRID	12		4.0	-5.0	2.0
GRID	13		7.0	0.0	2.0
GRID	14		-6.0	0.0	12.
GRID	15		-4.0	4.0	12.
GRID	16		-4.0	-4.0	12.
GRID	17		4.0	4.0	12.
GRID	18		4.0	-4.0	12.
GRID	19		6.0	0.0	12.0
GRID	26		-5.0	0.0	22.0
GRID	27		-4.0	3.0	22.0
GRID	28		-4.0	-3.0	22.0
GRID	2830		0.0	-3.0	22.0
GRID	1001		0.0	-6.5	22.0
GRID	29		4.0	3.0	22.0
GRID	30		4.0	-3.0	22.0
GRID	31		5.0	0.0	22.0
GRID	32		-4.0	10.0	22.0
GRID	3233		0.0	10.0	22.0
GRID	1003		0.0	6.5	22.0
GRID	33		4.0	10.0	22.0
GRID	34		-4.0	-10.0	22.0
GRID	35		4.0	-10.0	22.0
GRID	36		-4.0	3.0	24.0
GRID	37		-4.0	-3.0	24.0
GRID	38		4.0	3.0	24.0
GRID	39		4.0	-3.0	24.0
GRID	40		0.0	2.5	2.0
GRID	1002		0.0	0.0	2.0
GRID	42		0.0	5.0	-0.3
GRID	43		-2.0	0.0	-1.3
GRID	44		0.0	-1.667	-1.3
GRID	45		2.0	0.0	-1.3
GRID	46		-4.0	-5.0	-0.3
GRID	47		4.0	-5.0	-0.3
GRID	48		-26.0	0.0	-1.3
GRID	49		-21.00	0.0	-1.3
GRID	50		-16.0	0.0	-1.3
GRID	51		-11.0	0.0	-1.3
GRID	52		-6.0	0.0	-1.3
GRID	53		6.0	0.0	-1.3
GRID	54		11.0	0.0	-1.3
GRID	55		16.0	0.0	-1.3
GRID	56		21.00	0.0	-1.3
GRID	57		26.0	0.0	-1.3
GRID	100		0.0	0.0	0.0

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\$ ELEMENT CONNECTION DATA				LOCAL AXIS ORIENTATION VECTOR			
\$ ELEM#	PROP#	NODE	NODE	1.0	0.0	0.0	1
BAROR							
CBAR	1	200	1	2			
CBAR	2	200	1	3			
C3BAR	3	200	2	3			
CBAR	4	200	2	4	0.0	1.0	0.0
CBAR	5	200	3	4			

CBAR	6	200	4	5	0.0	1.0	0.0	1
CBAR	7	200	4	6				
CBAR	8	200	3	6	0.0	1.0	0.0	1
CBAR	9	200	5	6				
CBAR	10	200	5	7				
CBAR	11	200	6	7				
CBAR	12	200	1	8				
CBAR	13	200	2	9				
CBAR	14	200	3	10				
CBAR	15	200	5	11				
CBAR	16	200	6	12				
CBAR	17	200	7	13				
CBAR	18	200	3	8				
CBAR	19	200	2	8				
CBAR	21	200	4	9				
CBAR	22	200	4	11				
CBAR	24	200	5	13				
CBAR	25	200	6	13				
CBAR	30	200	8	9				
CBAR	31	200	8	10				
CBAR	32	200	9	910				
CBAR	232	200	910	10				
CBAR	33	200	9	40				
CBAR	34	200	910	40				
CBAR	35	200	11	40				
CBAR	36	200	1112	40				
CBAR	201	200	910	1112	0.0	1.0	0.0	1
CBAR	202	200	2	910				
CBAR	203	200	3	910				
CBAR	204	200	5	1112				
CBAR	205	200	6	1112				
CBAR	207	200	12	910				
CBAR	26	200	1112	3				
CBAR	27	200	6	10				
CBAR	37	200	9	11	0.0	1.0	0.0	1
CBAR	39	200	10	12	0.0	1.0	0.0	1
CBAR	39	200	11	1112				
CBAR	239	200	1112	12				
CBAR	40	200	11	13				
CBAR	41	200	12	13				
CBAR	42	300	14	15				
CBAR	43	300	14	16				
CSAR	44	300	16	15				
CSAR	45	300	17	18				
CBAR	46	300	17	19				
CBAR	47	300	18	19				
CBAR	54	300	24	27				
CBAR	55	300	26	28				
CBAR	56	300	27	28				
CBAR	57	300	29	30				
CSAR	58	300	29	31				
CBAR	59	300	30	31				
CBAR	60	300	27	29	0.0	1.0	0.0	1
CBAR	61	300	27	30				
CBAR	62	62	28	2830	0.0	1.0	0.0	1
CBAR	184	62	2830	30	0.0	1.0	0.0	1
CBAR	63	300	27	36				

CBAR	64	300	28	37				
CBAR	65	300	30	39				
CBAR	66	300	29	38				
CBAR	67	300	29	36				
CBAR	68	300	27	37				
CBAR	69	300	28	39				
CBAR	70	300	30	38				
CBAR	71	300	36	37				
CBAR	72	300	37	39	0.0	1.0	0.0	1
CBAR	73	300	39	38				
CBAR	74	300	36	38	0.0	1.0	0.0	1
CBAR	75	300	37	38				
CBAR	127	300	26	37				
CBAR	128	300	26	36				
CBAR	129	300	31	39				
CBAR	130	300	31	38				
CBAR	76	400	8	14				
CBAR	77	400	10	14				
CBAR	78	400	10	16				
CBAR	79	400	16	9				
CBAR	80	400	9	15				
CBAR	181	400	8	15				
CBAR	182	200	6	40				
CBAR	183	200	2	40				
CSAR	186	200	3	40				
CSAR	187	200	5	40				
CBAR	81	400	11	17				
CBAR	82	400	11	18				
CBAR	83	400	12	18				
CBAR	84	400	12	19				
CBAR	85	400	13	19				
CSAR	86	400	13	17				
CSAR	87	400	14	26				
CBAR	88	400	14	28				
CBAR	89	400	16	28				
CBAR	90	400	16	27				
CBAR	91	400	15	27				
CBAR	92	400	15	26				
CBAR	93	400	17	29				
CBAR	94	400	18	29				
CBAR	95	400	18	30				
CBAR	96	400	19	30				
CBAR	97	400	19	31				
CBAR	98	400	17	31				
CBAR	99	400	15	32				
CBAR	100	400	16	34				
CBAR	101	400	17	33				
CBAR	102	400	18	35				
CBAR	111	400	26	32				
CBAR	112	400	27	32				
CBAR	113	400	27	33				
CSAR	114	400	29	33				
CBAR	115	400	31	33				
CSAR	116	62	32	3233	0.0	1.0	0.0	1
CBAR	185	62	3233	33	0.0	1.0	0.0	1
CBAR	117	400	26	34				
CBAR	118	400	28	34				

	119	400	30	34					
CBAR	120	400	30	35					
CBAR	121	400	31	35					
CBAR	122	400	34	35	0.0	1.0	0.0	1	
CBAR	123	400	32	36					
CBAR	124	400	33	38					
CBAR	125	400	34	37					
CBAR	126	400	35	39					
CBAR	131	500	48	49	0.0	1.0	0.0	1	
CBAR	132	500	49	50	0.0	1.0	0.0	1	
CBAR	133	500	50	51	0.0	1.0	0.0	1	
CBAR	134	500	51	52	0.0	1.0	0.0	1	
CBAR	135	500	52	43	0.0	1.0	0.0	1	
CBAR	136	500	45	53	0.0	1.0	0.0	1	
CBAR	137	500	53	54	0.0	1.0	0.0	1	
CBAR	138	500	54	55	0.0	1.0	0.0	1	
CBAR	139	500	55	56	0.0	1.0	0.0	1	
CBAR	140	500	56	57	0.0	1.0	0.0	1	

\$ ISOLATOR SPRINGS

\$	ELEM#	K (N/M)	NODE A	DOF A	NODE B	DOF B	
\$	CELAS2	142	5.79E3	4	1	42	1
\$	CELAS2	143	5.79E3	4	2	42	2
\$	CELAS2	144	5.79E3	4	3	42	3
\$	CELAS2	145	5.79E3	3	1	46	1
\$	CELAS2	146	5.79E3	3	2	46	2
\$	CELAS2	147	5.79E3	3	3	46	3
\$	CELAS2	148	5.79E3	6	1	47	1
\$	CELAS2	149	5.79E3	6	2	47	2
\$	CELAS2	150	5.79E3	6	3	47	3

\$ MATERIAL PROPERTY DATA

\$	MAT #	E	NU	RHO
\$	MAT1	100	1.24E+11	0.3
\$	MAT1	200	1.24E+11	0.3
\$	MAT1	300	1.24E+11	0.3

\$ LUMPED MASS DATA

\$	CONN2	ELEM#	NODE#	MASS	+XXXX
\$	+XXXX	IXX	IYY	IZZ	
\$	CONN2	1001	1001	1000.	+1001
\$	+1001	4083.33	5333.33	9416.67	
\$	CONN2	1002	1002	800.	+4040
\$	+4040	1666.67	4266.67	5933.33	+1003
\$	CONN2	1003	1003	1200.	
\$	+1003	4900.	6400.	11300.	+1004
\$	CONN2	1004	1004	600.	

+1004	200.	800.	1000.				
\$							
\$	EQUIPMENT SECTION						
\$							
CONM2	544	44	3500.	+544			
+544	20611.	10500.	28777.				
\$							
\$	SOLAR PANELS						
\$							
CONM2	548	48	81.91	+548			
+548	270.0						
CONM2	550	50	163.82	+550			
+550	540.0						
CONM2	552	52	73.82	+552			
+552	270.0						
CONM2	553	53	73.82	+553			
+553	270.0						
CONM2	557	57	81.91	+557			
+555	540.0						
CONM2	555	55	163.82	+555			
+557	270.0						
\$							
\$	ADDITIONAL NON-STRUCTURAL MASS AT MIRROR SUPPORTS						
\$							
CONM2	501	27	69.5				
CONM2	502	28	6.74				
CONM2	503	29	69.5				
CONM2	504	30	6.74				
CONM2	505	32	6.74				
CONM2	506	33	6.74				
CONM2	507	34	69.5				
CONM2	508	35	69.5				
CONM2	509	9	67.4				
CONM2	510	10	67.4				
CONM2	511	11	67.4				
CONM2	512	12	67.4				
\$							
\$	BEAM SECTION PROPERTIES						
\$							
\$	PROP#	MAT#	AREA	IYY	IZZ	J	NSM
\$							
FBAR	200	100	6.250E-43.095E-63.095E-66.189E-6				
FBAR	300	100	3.133E-41.509E-61.509E-63.113E-6				
PBAR	400	100	3.919E-43.049E-63.049E-66.099E-6				
FBAR	500	100	9.407E-41.874E-51.874E-53.749E-5				
PBAR	62	300	6.082E-31.521E-31.521E033.041E-3	15.69			
\$							
\$							
\$	MULTI-POINT CONSTRAINT FOR X-AXIS LOS ERROR (NODE 100 DOF 1)						
\$							
MPC*	100		100	1	-1.0*1000000		
*1000000		34	2	-0.01855287570		*1000001	
*1000001			34	3	-0.14285714286*1000002		
*1000002		35	2	-0.0185528757		*1000003	
*1000003			35	3	-0.14285714286*1000004		
*1000004		2830	3	0.28571428572		*1000005	

```

*1000005          30      3   0.0      *1000006
*1000006          27      2   0.08065681999  *1000007
*1000007          27      3  -0.35489000795*1000008
*1000008          29      2   0.08065681999  *1000009
*1000009          29      3  -0.35489000795*1000010
*1000010          3233     3   0.70978001590  *1000011
*1000011          33      3   0.0      *1000012
*1000012          1002     4  -3.48423005566  *1000013
*1000013          11      2  -0.06210394429*1000014
*1000014          9       2  -0.06210394429

$      MULTI-POINT CONSTRAINT FOR Y-AXIS LOS ERROR (NODE 100 DOF 2)
$

MPC*  100          100      2   -1.0*2000000
*2000000          34      1  -0.03710575139  *2000001
*2000001          34      2  -0.04638218924*2000002
*2000002          34      3  -0.25000000000  *2000003
*2000003          35      3  0.25000000000  *2000004
*2000004          35      2   0.04638218924*2000004
*2000005          27      1   0.16131363998*2000006
*2000006          27      2  -0.06049261499  *2000007
*2000007          27      3  -0.62105751391*2000003
*2000008          29      2   0.06049261499  *2000009
*2000009          29      3   0.62105751391*2000010
*2000010          1002     5   3.48423005566  *2000011
*2000011          11      1  -0.12420788859*2000012
*2000012          11      2  -0.07762993037  *2000013
*2000013          9       2   0.07762993037*2000014
*2000014

$      MULTI-POINT CONSTRAINT EQUATION FOR DEFOCUS (NODE 100 DOF 3)
$

MPC*  100          100      3   -1.0*3000000
*3000000          34      3  -0.01912393776  *3000001
*3000001          35      3  -0.01912393776*3000002
*3000002          2830     3   0.12749291836  *3000003
*3000003          30      3   0.0      *3000004
*3000004          27      3   0.77803217347  *3000005
*3000005          29      3   0.77803217347*3000006
*3000006          3233     3  -0.46681930403  *3000007
*3000007          1002     3  -0.17849008571*3000008
*3000008          9       3   0.50000000000  *3000009
*3000009          11      3   0.50000000000*3000010
*3000010          40      3  -2.00000000000

$      RIGID BODY SUPPORT
$
SUPPORT,44,123456
ENDDATA

```

\*\*\*\*\* END OF MEMBER REV01      391 RECORDS \*\*\*\*\*

**APPENDIX B**

```

ID CRAPER,MCDEL2
SOL 3
CHKPNT YES
CEND
TITLE = ACROSS MODEL #2 - REVISION 3
SUBTITLE = VCOSS DESIGN MODEL
LABEL = 156 MODES AND FREQUENCIES
MPC = 100
METHOD = 600
DISP = ALL
$ESE = ALL
BEGIN BULK
PARAM,USETPRT,1
PARAM GROPNT 0
EIGR 600 GIV 200 +10
+10 MASS
$
$ KINEMATIC MOUNT: TERTIARY MIRROR
$ RBE1,1003,27,123,29,23,3233,3,,+RB31
+RB31,UM,1003,123456
$ KINEMATIC MOUNT: PRIMARY MIRROR
$ RBE1,1001,34,123,35,23,2830,3,,+RB11
+RB11,UM,1001,123456
$ KINEMATIC MOUNT: FOCAL PLANE
$ RBE1,1004,11,123,9,23,40,3,,+RB41
+RB41,UM,1004,123456
$ KINEMATIC MOUNT: SECONDARY MIRROR
$ RBE1,1002,910,123,1112,23,40,3,,+RB21
+RB21,UM,1002,123456
$ RIGID EQUIPMENT SECTION
$ RBE2 141 44 123456 42 43 45 46 47
$ NODE POINT LOCATIONS
$ NODE# X (M) Y (M) Z (M)
$ GRID 1 -7.0 0.0 0.0
GRID 2 -4.0 5.0 0.0
GRID 3 -4.0 -5.0 0.0
GRID 4 0.0 5.0 0.0
GRID 5 4.0 5.0 0.0
GRID 6 4.0 -5.0 0.0
GRID 7 7.0 0.0 0.0
GRID 8 -7.0 0.0 2.0
GRID 9 -4.0 5.0 2.0
GRID 1004 0.0 4.0 2.0
GRID 10 -4.0 -5.0 2.0
GRID 11 4.0 5.0 2.0
GRID 12 4.0 -5.0 2.0

```

GRID	910		-4.0	-2.5	2.0		
GRID	1112		4.0	-2.5	2.0		
GRID	13		7.0	0.0	2.0		
GRID	14		-6.0	0.0	12.		
GRID	15		-4.0	4.0	12.		
GRID	16		-4.0	-4.0	12.		
GRID	17		4.0	4.0	12.		
GRID	18		4.0	-4.0	12.		
GRID	19		6.0	0.0	12.0		
GRID	26		-5.0	0.0	22.0		
GRID	27		-4.0	3.0	22.0		
GRID	28		-4.0	-3.0	22.0		
GRID	2830		0.0	-3.0	22.0		
GRID	1001		0.0	-6.5	22.0		
GRID	29		4.0	3.0	22.0		
GRID	30		4.0	-3.0	22.0		
GRID	31		5.0	0.0	22.0		
GRID	32		-4.0	10.0	22.0		
GRID	3233		0.0	10.0	22.0		
GRID	1003		0.0	6.5	22.0		
GRID	33		4.0	10.0	22.0		
GRID	34		-4.0	-10.0	22.0		
GRID	35		4.0	-10.0	22.0		
GRID	36		-4.0	3.0	24.0		
GRID	37		-4.0	-3.0	24.0		
GRID	38		4.0	3.0	24.0		
GRID	39		4.0	-3.0	24.0		
GRID	40		0.0	2.5	2.0		
GRID	1002		0.0	0.0	2.0		
GRID	42		0.0	5.0	-0.3		
GRID	43		-2.0	0.0	-1.3		
GRID	44		0.0	-1.667	-1.3		
GRID	45		2.0	0.0	-1.3		
GRID	46		-4.0	-5.0	-0.3		
GRID	47		4.0	-5.0	-0.3		
GRID	48		-26.0	0.0	-1.3		
GRID	49		-21.00	0.0	-1.3		
GRID	50		-16.0	0.0	-1.3		
GRID	51		-11.0	0.0	-1.3		
GRID	52		-6.0	0.0	-1.3		
GRID	53		6.0	0.0	-1.3		
GRID	54		11.0	0.0	-1.3		
GRID	55		16.0	0.0	-1.3		
GRID	56		21.00	0.0	-1.3		
GRID	57		26.0	0.0	-1.3		
GRID	100		0.0	0.0	0.0	456	
\$							
\$	ELEMENT CONNECTION DATA						
\$							
\$	ELEM#	PROP#	NODE A	NODE B	LOCAL AXIS ORIENTATION VECTOR		
\$							
\$	BAROR						
CBAR	1	1	1	2	1.0	0.0	0.0
CSBAR	2	2	1	3			
CBAR	3	3	2	3			
CSBAR	4	4	2	4	0.0	1.0	0.0
							1

CBAR	5	5	3	4				
CBAR	6	6	4	5	0.0	1.0	0.0	1
CBAR	7	7	4	6				
CBAR	8	8	3	6	0.0	1.0	0.0	1
CBAR	9	9	5	6				
CBAR	10	10	5	7				
CBAR	11	11	6	7				
CBAR	12	12	1	8				
CBAR	13	13	2	9				
CBAR	14	14	3	10				
CBAR	15	15	5	11				
CBAR	16	16	6	12				
CBAR	17	17	7	13				
CBAR	18	18	3	8				
CBAR	19	19	2	8				
CBAR	21	21	4	9				
CBAR	22	22	4	11				
CBAR	24	24	5	13				
CBAR	25	25	6	13				
CBAR	26	26	1112	3				
CBAR	27	27	6	10				
CBAR	30	30	8	9				
CBAR	31	31	8	10				
CBAR	32	32	9	910				
CBAR	232	232	910	10				
CBAR	33	33	9	40				
CBAR	34	34	910	40				
CBAR	35	35	11	40				
CBAR	36	36	1112	40				
CBAR	37	37	9	11	0.0	1.0	0.0	1
CBAR	38	38	10	12	0.0	1.0	0.0	1
CBAR	39	39	11	1112				
CBAR	239	239	1112	12				
CBAR	201	201	910	1112	0.0	1.0	0.0	1
CBAR	202	202	2	910				
CBAR	203	203	3	910				
CBAR	204	204	5	1112				
CBAR	205	205	6	1112				
CBAR	207	207	12	910				
CBAR	40	40	11	13				
CBAR	41	41	12	13				
CBAR	42	42	14	15				
CBAR	43	43	14	16				
CBAR	44	44	16	15				
CBAR	45	45	17	18				
CBAR	46	46	17	19				
CBAR	47	47	18	19				
CBAR	54	54	26	27				
CBAR	55	55	26	28				
CBAR	56	56	27	28				
CBAR	57	57	29	30				
CBAR	58	58	29	31				
CBAR	59	59	30	31				
CBAR	60	60	27	29	0.0	1.0	0.0	1
CBAR	61	61	27	30				
CBAR	62	62	28	2830	0.0	1.0	0.0	1
CBAR	184	184	2830	30	0.0	1.0	0.0	1

CBAR	63	63	27	36				
CBAR	64	64	28	37				
CBAR	65	65	30	39				
CBAR	66	66	29	38				
CBAR	67	67	29	36				
CBAR	68	68	27	37				
CBAR	69	69	28	39				
CBAR	70	70	30	38				
CBAR	71	71	36	37				
CBAR	72	72	37	39	0.0	1.0	0.0	1
CBAR	73	73	39	38				
CBAR	74	74	36	38	0.0	1.0	0.0	1
CBAR	75	75	37	38				
CBAR	127	127	26	37				
CBAR	128	128	26	36				
CBAR	129	129	31	39				
CBAR	130	130	31	38				
CBAR	76	76	8	14				
CBAR	77	77	10	14				
CBAR	78	78	10	16				
CBAR	79	79	16	9				
CBAR	80	80	9	15				
CBAR	181	181	8	15				
CBAR	182	182	6	40				
CBAR	183	183	2	40				
CBAR	186	165	3	40				
CBAR	187	187	5	40				
CBAR	81	81	11	17				
CBAR	82	82	11	18				
CBAR	83	83	12	18				
CBAR	84	84	12	19				
CBAR	85	85	13	19				
CBAR	86	86	13	17				
CBAR	87	87	14	26				
CBAR	88	88	14	28				
CBAR	89	89	16	28				
CBAR	90	90	16	27				
CBAR	91	91	15	27				
CBAR	92	92	15	26				
CBAR	93	93	17	29				
CBAR	94	94	18	29				
CBAR	95	95	18	30				
CBAR	96	96	19	30				
CBAR	97	97	19	31				
CBAR	98	98	17	31				
CBAR	99	99	15	32				
CBAR	100	100	16	34				
CBAR	101	101	17	33				
CBAR	102	102	18	35				
CBAR	111	111	26	32				
CBAR	112	112	27	32				
CBAR	113	113	27	33				
CBAR	114	114	29	33				
CBAR	115	115	31	33				
CBAR	116	116	32	3233	0.0	1.0	0.0	1
CBAR	185	185	3233	33	0.0	1.0	0.0	1
CBAR	117	117	26	34				

CBAR	118	118	28	34				
CBAR	119	119	30	34				
CSAR	120	120	30	35				
CBAR	121	121	31	35				
CBAR	122	122	34	35	0.0	1.0	0.0	1
CBAR	123	123	32	36				
CBAR	124	124	33	38				
CBAR	125	125	34	37				
CBAR	126	126	35	39				
CBAR	131	131	48	49	0.0	1.0	0.0	1
CBAR	132	132	49	50	0.0	1.0	0.0	1
CBAR	133	133	50	51	0.0	1.0	0.0	1
CBAR	134	134	51	52	0.0	1.0	0.0	1
CBAR	135	135	52	43	0.0	1.0	0.0	1
CBAR	136	136	45	53	0.0	1.0	0.0	1
CBAR	137	137	53	54	0.0	1.0	0.0	1
CBAR	138	138	54	55	0.0	1.0	0.0	1
CBAR	139	139	55	56	0.0	1.0	0.0	1
CBAR	140	140	56	57	0.0	1.0	0.0	1
\$								
\$	ISOLATOR SPRINGS							
\$								
\$	ELEM#	K (N/M)	NODE A	DOF A	NODE B	DOF B		
\$								
CELAS2	142	5.79E3	4	1	42	1		
CELAS2	143	5.79E3	4	2	42	2		
CELAS2	144	5.79E3	4	3	42	3		
CELAS2	145	5.79E3	3	1	46	1		
CELAS2	146	5.79E3	3	2	46	2		
CELAS2	147	5.79E3	3	3	46	3		
CELAS2	148	5.79E3	6	1	47	1		
CELAS2	149	5.79E3	6	2	47	2		
CELAS2	150	5.79E3	6	3	47	3		
\$								
\$	MATERIAL PROPERTY DATA							
\$								
\$	, MAT#	E		NU	RHO			
\$								
MAT1	100	1.24E+11		0.3	1720.			
MAT1	200	1.24E+11		0.3	1720.			
MAT1	300	1.24E+11		0.3	2579.70			
\$								
\$	LUMPED MASS DATA							
\$								
\$CONM2	ELEM#	NODE#		MASS			+XXXX	
\$+XXXX	IXX		IYY		IZZ			
\$								
\$	MIRRORS							
\$								
CONM2	1001	1001		1000.			+1001	
+1001	4083.33		5333.33		9416.67			
CONM2	1002	1002		800.			+4040	
+4040	1666.67		4266.67		5933.33			
CONM2	1003	1003		1200.			+1003	
+1003	4900.		6400.		11300.			
CONM2	1004	1004		600.			+1004	

+1004	200.	600.	1000.	
\$				
\$	EQUIPMENT SECTION			
\$				
CONM2	544	44	3500.	+544
+544	29611.	10500.	28777.	
\$				
\$	SOLAR PANELS			
\$				
CONM2	548	48	81.91	+548
+548	270.0			
CONM2	550	50	163.82	+550
+550	540.0			
CONM2	552	52	73.82	+552
+552	270.0			
CONM2	553	53	73.82	+553
+553	270.0			
CONM2	557	57	81.91	+557
+555	540.0			
CONM2	555	55	163.82	+555
+557	270.0			
\$				
\$	ADDITIONAL NON-STRUCTURAL MASS AT MIRROR SUPPORTS			
\$				
CONM2	501	27	69.5	
CONM2	502	28	6.74	
CONM2	503	29	69.5	
CONM2	504	30	6.74	
CONM2	505	32	6.74	
CONM2	506	33	6.74	
CONM2	507	34	69.5	
CONM2	508	35	69.5	
CONM2	509	9	67.4	
CONM2	510	10	67.4	
CONM2	511	11	67.4	
CONM2	512	12	67.4	
\$				
\$	BEAM SECTION PROPERTIES			
\$				
\$PBAR		PROP#	MAT#	
\$*XXXXXX	I22	J		AREA
PBAR*		1	100	0.678583E-04
*	1	0.439721E-07	0.879442E-07	0.439721E-07*
PBAR*		2	100	0.678583E-04
*	2	0.439721E-07	0.879442E-07	0.439721E-07*
PBAR*		3	100	0.255098E-03
*	3	0.621422E-06	0.124284E-05	0.621422E-06*
PBAR*		4	100	0.678583E-04
*	4	0.439721E-07	0.879443E-07	0.439721E-07*
PBAR*		5	100	0.343532E-03
*	5	0.112695E-05	0.225391E-05	0.112695E-05*
PBAR*		6	100	0.678583E-04
*	6	0.439721E-07	0.879443E-07	0.439721E-07*
PBAR*		7	100	0.343532E-03
*	7	0.112695E-05	0.225391E-05	0.112695E-05*
PBAR*		8	100	0.104152E-03
				0.103587E-06*

*	8	0.103587E-06	0.207174E-06			
PBAR*	9	0.621422E-06	0.124284E-05	0.255098E-03	0.621422E-06*	9
*	10	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	10
PBAR*	11	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	11
*	12	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	12
PBAR*	13	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	13
*	14	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	14
PBAR*	15	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	15
*	16	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	16
PBAR*	17	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	17
*	18	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	18
PBAR*	19	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	19
*	21	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	21
PBAR*	22	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	22
*	24	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	24
PBAR*	25	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	25
*	26	0.188490E-06	0.376979E-06	0.140494E-03	0.188490E-06*	26
PBAR*	27	0.188490E-06	0.376979E-06	0.140494E-03	0.188490E-06*	27
*	30	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	30
PBAR*	31	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	31
*	32	0.618177E-07	0.123635E-06	0.804583E-04	0.618177E-07*	32
FBAR*	33	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	33
PBAR*	34	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	34
*	35	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	35
PBAR*	36	0.439721E-07	0.879443E-07	0.678583E-04	0.439721E-07*	36
*	37	0.103587E-06	0.207174E-06	0.104152E-03	0.103587E-06*	37
PBAR*	38	0.103587E-06	0.207174E-06	0.104152E-03	0.103587E-06*	38
*	39	0.618177E-07	0.123635E-06	0.804583E-04	0.618177E-07*	39
PBAR*	40	0.439721E-07	0.879442E-07	0.678583E-04	0.439721E-07*	40

PBAR*		41	100	0.678583E-04	0.439721E-07*	41
*	41	0.439721E-07	0.879442E-07			
PBAR*		42	100	0.678583E-04	0.439721E-07*	42
*	42	0.439721E-07	0.879443E-07			
PBAR*		43	100	0.678583E-04	0.439721E-07*	43
*	43	0.439721E-07	0.879443E-07			
PBAR*		44	100	0.104152E-03	0.103587E-06*	44
*	44	0.103587E-06	0.207174E-06			
PBAR*		45	100	0.104152E-03	0.103587E-06*	45
*	45	0.103587E-06	0.207174E-06			
PBAR*		46	100	0.678583E-04	0.439721E-07*	46
*	46	0.439721E-07	0.879443E-07			
PBAR*		47	100	0.678583E-04	0.439721E-07*	47
*	47	0.439721E-07	0.879443E-07			
PBAR*		54	100	0.678583E-04	0.439721E-07*	54
*	54	0.439721E-07	0.879442E-07			
PBAR*		55	100	0.678583E-04	0.439721E-07*	55
*	55	0.439721E-07	0.879442E-07			
PBAR*		56	100	0.678583E-04	0.439721E-07*	56
*	56	0.439721E-07	0.879442E-07			
PBAR*		57	100	0.678583E-04	0.439721E-07*	57
*	57	0.439721E-07	0.879442E-07			
PBAR*		58	100	0.678583E-04	0.439721E-07*	58
*	58	0.439721E-07	0.879442E-07			
PBAR*		59	100	0.678583E-04	0.439721E-07*	59
*	59	0.439721E-07	0.879442E-07			
PBAR*		60	100	0.104152E-03	0.103587E-06*	60
*	60	0.103587E-06	0.207174E-06			
PBAR*		61	100	0.255098E-03	0.621422E-06*	61
*	61	0.621422E-06	0.124284E-05			
PBAR*		62	300	0.0060821	0.0015205 *	62
*	62	0.0015205	0.0030410			
PBAR*		63	100	0.678583E-04	0.439721E-07*	63
*	63	0.439721E-07	0.879443E-07			
PBAR*		64	100	0.678583E-04	0.439721E-07*	64
*	64	0.439721E-07	0.879443E-07			
PBAR*		65	100	0.678583E-04	0.439721E-07*	65
*	65	0.439721E-07	0.879443E-07			
PBAR*		66	100	0.678583E-04	0.439721E-07*	66
*	66	0.439721E-07	0.879443E-07			
PBAR*		67	100	0.117640E-03	0.132154E-06*	67
*	67	0.132154E-06	0.264308E-06			
PBAR*		68	100	0.678583E-04	0.439721E-07*	68
*	68	0.439721E-07	0.879442E-07			
PBAR*		69	100	0.117640E-03	0.132154E-06*	69
*	69	0.132154E-06	0.264308E-06			
PBAR*		70	100	0.678583E-04	0.439721E-07*	70
*	70	0.439721E-07	0.879442E-07			
PBAR*		71	100	0.678583E-04	0.439721E-07*	71
*	71	0.439721E-07	0.879442E-07			
PBAR*		72	100	0.104152E-03	0.103587E-06*	72
*	72	0.103587E-06	0.207174E-06			
PBAR*		73	100	0.678583E-04	0.439721E-07*	73
*	73	0.439721E-07	0.879442E-07			
PBAR*		74	100	0.104152E-03	0.103587E-06*	74
*	74	0.103587E-06	0.207174E-06			
PBAR*		75	100	0.255098E-03	0.621422E-06*	75







```

*2000003          35      2   0.04638218924*2000004
*2000004          35      3   0.2500000000           *2000005
*2000005          27      1   0.16131363998*2000006
*2000006          27      2   -0.06049261499        *2000007
*2000007          27      3   -0.62105751391*2000008
*2000008          29      2   0.06049261499        *2000009
*2000009          29      3   0.62105751391*2000010
*2000010         1002     5   3.48423005566       *2000011
*2000011          11      1   -0.12420788859*2000012
*2000012          11      2   -0.07762993037       *2000013
*2000013          9       2   0.07762993037*2000014
*2000014          9       2   0.07762993037*2000014
$
$      MULTI-POINT CONSTRAINT EQUATION FOR DEFOCUS (NODE 100 DOF 3)
$
MPC#    100          100      3   -1.0*3000000
*3000000          34      3   -0.01912393776       *3000001
*3000001          35      3   -0.01912393776*3000002
*3000002         2830     3   0.12749291836       *3000003
*3000003          30      3   0.0                   *3000004
*3000004          27      3   0.77803217347       *3000005
*3000005          29      3   0.77803217347*3000006
*3000006         3233     3   -0.46681930408       *3000007
*3000007         1002     3   -0.17849008571*3000008
*3000008          9       3   0.500000000000       *3000009
*3000009          11      3   0.500000000000*3000010
*3000010          40      3   -2.000000000000
$
$      RIGID BODY SUPPORT
$
SUPORT,44,123456
ENDDATA

```

\*\*\*\*\* END OF MEMBER REV03 659 RECORDS \*\*\*\*\*

**APPENDIX C**

```

ID DRAPER,MODEL2
SOL 3
CHKPNT YES
TIME 10
CEND
TITLE = ACOSS MODEL #2
SUBTITLE = MODIFIED MODEL - REVISION 4
LASEL = VCOSS STIFFNESS MODEL
MPC = 100
METHOD = 600
DISP = ALL
ESE =ALL
BEGIN BULK
PARAM,USETPRT,1
PARAM GRDPNT 0
EIGR 600 GIV
+10 MASS
$
$ KINEMATIC MOUNT: TERTIARY MIRROR
$ RBE1,1003,27,123,29,23,3233,3,,+RB31
+RB31,UM,1003,123456
$ KINEMATIC MOUNT: PRIMARY MIRROR
$ RBE1,1001,34,123,35,23,2830,3,,+RB11
+RB11,UM,1001,123456
$ KINEMATIC MOUNT: FOCAL PLANE
$ RBE1,1004,11,123,9,23,40,3,,+RB41
+RB41,UM,1004,123456
$ KINEMATIC MOUNT: SECONDARY MIRROR
$ RBE1,1002,910,123,1112,23,40,3,,+RB21
+RB21,UM,1002,123456
$ RIGID EQUIPMENT SECTION
$ RBE2 141 44 123456 42 43 45 46 47
$ NODE POINT LOCATIONS
$ NODE# X (M) Y (M) Z (M)
$ GRID 1 -7.0 0.0 0.0
GRID 2 -4.0 5.0 0.0
GRID 3 -4.0 -5.0 0.0
GRID 4 0.0 5.0 0.0
GRID 5 4.0 5.0 0.0
GRID 6 4.0 -5.0 0.0
GRID 7 7.0 0.0 0.0
GRID 8 -7.0 0.0 2.0
GRID 9 -4.0 5.0 2.0
GRID 1004 0.0 4.0 2.0
GRID 10 -4.0 -5.0 2.0
GRID 910 -4.0 -2.5 2.0

```

GRID	1112	4.0	-2.5	2.0
GRID	11	4.0	5.0	2.0
GRID	12	4.0	-5.0	2.0
GRID	13	7.0	0.0	2.0
GRID	14	-6.0	0.0	12.
GRID	15	-4.0	4.0	12.
GRID	16	-4.0	-4.0	12.
GRID	17	4.0	4.0	12.
GRID	18	4.0	-4.0	12.
GRID	19	6.0	0.0	12.0
GRID	26	-5.0	0.0	22.0
GRID	27	-4.0	3.0	22.0
GRID	28	-4.0	-3.0	22.0
GRID	2830	0.0	-3.0	22.0
GRID	1001	0.0	-6.5	22.0
GRID	29	4.0	3.0	22.0
GRID	30	4.0	-3.0	22.0
GRID	31	5.0	0.0	22.0
GRID	32	-4.0	10.0	22.0
GRID	3233	0.0	10.0	22.0
GRID	1003	0.0	6.5	22.0
GRID	33	4.0	10.0	22.0
GRID	34	-4.0	-10.0	22.0
GRID	35	4.0	-10.0	22.0
GRID	36	-4.0	3.0	24.0
GRID	37	-4.0	-3.0	24.0
GRID	38	4.0	3.0	24.0
GRID	39	4.0	-3.0	24.0
GRID	40	0.0	2.5	2.0
GRID	1002	0.0	0.0	2.0
GRID	42	0.0	5.0	-0.3
GRID	43	-2.0	0.0	-1.3
GRID	44	0.0	-1.667	-1.3
GRID	45	2.0	0.0	-1.3
GRID	46	-4.0	-5.0	-0.3
GRID	47	4.0	-5.0	-0.3
GRID	48	-26.0	0.0	-1.3
GRID	49	-21.00	0.0	-1.3
GRID	50	-16.0	0.0	-1.3
GRID	51	-11.0	0.0	-1.3
GRID	52	-6.0	0.0	-1.3
GRID	53	6.0	0.0	-1.3
GRID	54	11.0	0.0	-1.3
GRID	55	16.0	0.0	-1.3
GRID	56	21.00	0.0	-1.3
GRID	57	26.0	0.0	-1.3
GRID	100	0.0	0.0	0.0

456

ELEMENT CONNECTION DATA				
\$	ELEM#	PROP#	NODE A	NODE B
\$				LOCAL AXIS ORIENTATION VECTOR
\$				1.0 0.0 0.0 1
BARCR				
CBAR	1	1	1	2
CBAR	2	2	1	3
CBAR	3	3	2	3





CBAR 117 117 26 34  
 CBAR 118 118 28 34  
 CBAR 119 119 30 34  
 CBAR 120 120 30 35  
 CBAR 121 121 31 35  
 CBAR 122 122 34 35 0.0 1.0 0.0 1  
 CBAR 123 123 32 36  
 CBAR 124 124 33 38  
 CBAR 125 125 34 37  
 CBAR 126 126 35 39  
 CBAR 131 131 48 49 0.0 1.0 0.0 1  
 CBAR 132 132 49 50 0.0 1.0 0.0 1  
 CBAR 133 133 50 51 0.0 1.0 0.0 1  
 CBAR 134 134 51 52 0.0 1.0 0.0 1  
 CBAR 135 135 52 43 0.0 1.0 0.0 1  
 CBAR 136 136 45 53 0.0 1.0 0.0 1  
 CBAR 137 137 53 54 0.0 1.0 0.0 1  
 CBAR 138 138 54 55 0.0 1.0 0.0 1  
 CBAR 139 139 55 56 0.0 1.0 0.0 1  
 CBAR 140 140 56 57 0.0 1.0 0.0 1  
 \$  
 \$ ISOLATOR SPRINGS  
 \$  
 \$ ELEM# K NODE DOF NODE DOF  
 \$ (N/M) A A B B  
 \$  
 CELAS2 142 5.79E3 4 1 42 1  
 CELAS2 143 5.79E3 4 2 42 2  
 CELAS2 144 5.79E3 4 3 42 3  
 CELAS2 145 5.79E3 3 1 46 1  
 CELAS2 146 5.79E3 3 2 46 2  
 CELAS2 147 5.79E3 3 3 46 3  
 CELAS2 148 5.79E3 6 1 47 1  
 CELAS2 149 5.79E3 6 2 47 2  
 CELAS2 150 5.79E3 6 3 47 3  
 \$  
 \$ MATERIAL PROPERTY DATA  
 \$  
 \$ MAT# E NU RHO  
 \$  
 MAT1 100 1.24E+11 0.3 1720.  
 MAT1 200 1.24E+11 0.3 1720.  
 MAT1 300 1.24E+11 0.3  
 \$  
 \$ LUMPED MASS DATA  
 \$  
 \$ CONM2 ELEM# NODE# MASS +XXXX  
 \$ +XXXX IXX IYY IZZ  
 \$  
 \$ MIRRORS  
 \$  
 CONM2 1001 1001 1000. +1001  
 +1001 4083.33 5333.33 9416.67  
 \$  
 CONM2 1002 1002 800. +4040  
 +4040 1666.67 4266.67 5933.33  
 \$

CONM2 1003 1003 1200. +1003  
 +1003 4900. 6400. 11300.  
 \$ CONM2 1004 1004 600. +1004  
 +1004 200. 600. 1000.  
 \$ EQUIPMENT SECTION  
 \$ CONM2 544 44 3500. +544  
 +544 20611. 10500. 28777.  
 \$ SOLAR PANELS  
 \$ CONM2 548 48 61.91 +548  
 +548 270.0  
 CONM2 550 50 163.82 +550  
 +550 540.0  
 CONM2 552 52 73.82 +552  
 +552 270.0  
 CONM2 553 53 73.82 +553  
 +553 270.0  
 CONM2 557 57 61.91 +557  
 +555 540.0  
 CONM2 555 55 163.82 +555  
 +557 270.0  
 \$ ADDITIONAL NON-STRUCTURAL MASS AT MIRROR SUPPORTS  
 \$ CONM2 501 27 69.5  
 CONM2 502 28 6.74  
 CONM2 503 29 69.5  
 CONM2 504 30 6.74  
 CONM2 505 32 6.74  
 CONM2 506 33 6.74  
 CONM2 507 34 69.5  
 CONM2 508 35 69.5  
 CONM2 509 9 67.4  
 CONM2 510 10 67.4  
 CONM2 511 11 67.4  
 CONM2 512 12 67.4  
 \$ BEAM SECTION PROPERTIES  
 \$  
 \$FBAR PROP# MAT# AREA ILL \*XXXXXXX  
 \$\*XXXXXXX I22 J  
 \$  
 PBAR\* 1100 0.275119E-02 0.301163E-04\* 1  
 \* 1 0.301163E-04 0.602325E-04  
 PBAR\* 2100 0.275119E-02 0.301163E-04\* 2  
 \* 2 0.301163E-04 0.602325E-04  
 PBAR\* 3100 0.471826E-02 0.885774E-04\* 3  
 \* 3 0.885774E-04 0.177155E-03  
 PBAR\* 4100 0.188730E-02 0.141724E-04\* 4  
 \* 4 0.141724E-04 0.283447E-04  
 PBAR\* 5100 0.503172E-02 0.102750E-03\* 5  
 \* 5 0.102750E-03 0.205499E-03  
 FBAR\* 6100 0.188730E-02 0.141724E-04\* 6



PBAR*		39100		0.353969E-02	0.493248E-04*	39
*	39	0.498248E-04	0.996495E-04			
PBAR*		40100		0.275119E-02	0.301163E-04*	40
*	40	0.301163E-04	0.602325E-04			
PBAR*		41100		0.275119E-02	0.301163E-04*	41
*	41	0.301163E-04	0.602325E-04			
PBAR*		42100		0.211007E-02	0.177155E-04*	42
*	42	0.177155E-04	0.354309E-04			
PBAR*		43100		0.211007E-02	0.177155E-04*	43
*	43	0.177155E-04	0.354309E-04			
PBAR*		44100		0.377461E-02	0.566895E-04*	44
*	44	0.566895E-04	0.113379E-03			
PBAR*		45100		0.377461E-02	0.566895E-04*	45
*	45	0.566895E-04	0.113379E-03			
PBAR*		46100		0.211007E-02	0.177155E-04*	46
*	46	0.177155E-04	0.354309E-04			
PBAR*		47100		0.211007E-02	0.177155E-04*	47
*	47	0.177155E-04	0.354309E-04			
PBAR*		54100		0.149204E-02	0.885772E-05*	54
*	54	0.885772E-05	0.177154E-04			
PBAR*		55100		0.149204E-02	0.885772E-05*	55
*	55	0.885772E-05	0.177154E-04			
PBAR*		56100		0.283095E-02	0.318878E-04*	56
*	56	0.318878E-04	0.637757E-04			
PBAR*		57100		0.283095E-02	0.318878E-04*	57
*	57	0.318878E-04	0.637757E-04			
PBAR*		58100		0.149204E-02	0.885772E-05*	58
*	58	0.885772E-05	0.177154E-04			
PBAR*		59100		0.149204E-02	0.885772E-05*	59
*	59	0.885772E-05	0.177154E-04			
PBAR*		60100		0.377461E-02	0.566895E-04*	60
*	60	0.566895E-04	0.113379E-03			
PBAR*		61100		0.471826E-02	0.885774E-04*	61
*	61	0.885774E-04	0.177155E-03			
PBAR*		63100		0.943651E-03	0.354309E-05*	63
*	63	0.354309E-05	0.708619E-05			
PBAR*		64100		0.943651E-03	0.354309E-05*	64
*	64	0.354309E-05	0.708619E-05			
PBAR*		65100		0.943651E-03	0.354309E-05*	65
*	65	0.354309E-05	0.708619E-05			
PBAR*		66100		0.943651E-03	0.354309E-05*	66
*	66	0.354309E-05	0.708619E-05			
PBAR*		67100		0.389078E-02	0.602326E-04*	67
*	67	0.602326E-04	0.120465E-03			
PBAR*		68100		0.298409E-02	0.354310E-04*	68
*	68	0.354310E-04	0.708619E-04			
PBAR*		69100		0.389078E-02	0.602326E-04*	69
*	69	0.602326E-04	0.120465E-03			
PBAR*		70100		0.298409E-02	0.354310E-04*	70
*	70	0.354310E-04	0.708619E-04			
PBAR*		71100		0.283095E-02	0.318878E-04*	71
*	71	0.318878E-04	0.637757E-04			
PBAR*		72100		0.377461E-02	0.566895E-04*	72
*	72	0.566895E-04	0.113379E-03			
PBAR*		73100		0.283095E-02	0.318878E-04*	73
*	73	0.318878E-04	0.637757E-04			
PBAR*		74100		0.377461E-02	0.566895E-04*	74

*	74	0.566895E-04	0.113379E-03			
PBAR*		75100		0.471826E-02	0.885774E-04*	75
*	75	0.885774E-04	0.177155E-03			
PBAR*		76100		0.474179E-02	0.894631E-04*	76
*	76	0.894631E-04	0.178926E-03			
PBAR*		77100		0.535890E-02	0.114265E-03*	77
*	77	0.114265E-03	0.228529E-03			
PBAR*		78100		0.474179E-02	0.894631E-04*	78
*	78	0.894631E-04	0.178926E-03			
PBAR*		79100		0.634776E-02	0.160325E-03*	79
*	79	0.160325E-03	0.320650E-03			
PBAR*		80100		0.474179E-02	0.894631E-04*	80
*	80	0.894631E-04	0.178926E-03			
PBAR*		81100		0.474179E-02	0.894631E-04*	81
*	81	0.894631E-04	0.178926E-03			
PBAR*		82100		0.634776E-02	0.160325E-03*	82
*	82	0.160325E-03	0.320650E-03			
PBAR*		83100		0.474179E-02	0.894631E-04*	83
*	83	0.894631E-04	0.178926E-03			
PBAR*		84100		0.535890E-02	0.114265E-03*	84
*	84	0.114265E-03	0.228529E-03			
PBAR*		85100		0.474179E-02	0.894631E-04*	85
*	85	0.894631E-04	0.178926E-03			
PBAR*		86100		0.527517E-02	0.110722E-03*	86
*	86	0.110722E-03	0.221444E-03			
PBAR*		87100		0.474179E-02	0.894631E-04*	87
*	87	0.894631E-04	0.178926E-03			
PBAR*		88100		0.501557E-02	0.100092E-03*	88
*	88	0.100092E-03	0.200185E-03			
PBAR*		89100		0.474179E-02	0.894631E-04*	89
*	89	0.894631E-04	0.178926E-03			
PBAR*		90100		0.575936E-02	0.131980E-03*	90
*	90	0.131980E-03	0.263960E-03			
PBAR*		91100		0.474179E-02	0.894631E-04*	91
*	91	0.894631E-04	0.178926E-03			
PBAR*		92100		0.510358E-02	0.103636E-03*	92
*	92	0.103636E-03	0.207271E-03			
PBAR*		93100		0.474179E-02	0.894631E-04*	93
*	93	0.894631E-04	0.178926E-03			
PBAR*		94100		0.575936E-02	0.131980E-03*	94
*	94	0.131980E-03	0.263960E-03			
PBAR*		95100		0.474179E-02	0.894631E-04*	95
*	95	0.894631E-04	0.178926E-03			
PBAR*		96100		0.501557E-02	0.100092E-03*	96
*	96	0.100092E-03	0.200185E-03			
PBAR*		97100		0.474179E-02	0.894631E-04*	97
*	97	0.894631E-04	0.178926E-03			
PBAR*		98100		0.510358E-02	0.103636E-03*	98
*	98	0.103636E-03	0.207271E-03			
PBAR*		99100		0.550238E-02	0.120465E-03*	99
*	99	0.120465E-03	0.240930E-03			
PBAR*		100100		0.550238E-02	0.120465E-03*	100
*	100	0.120465E-03	0.240930E-03			
PBAR*		101100		0.550238E-02	0.120465E-03*	101
*	101	0.120465E-03	0.240930E-03			
PBAR*		102100		0.550238E-02	0.120465E-03*	102
*	102	0.120465E-03	0.240930E-03			

PSAR*	111100		0.474179E-02	0.894631E-04*	111
* 111	0.894631E-04	0.178926E-03			
PSAR*	112100		0.330278E-02	0.434029E-04*	112
* 112	0.434029E-04	0.868059E-04			
PSAR*	113100		0.501557E-02	0.100092E-03*	113
* 113	0.100092E-03	0.200185E-03			
PSAR*	114100		0.330278E-02	0.434029E-04*	114
* 114	0.434029E-04	0.868059E-04			
PSAR*	115100		0.474179E-02	0.894631E-04*	115
* 115	0.894631E-04	0.178926E-03			
PSAR*	117100		0.474179E-02	0.894631E-04*	117
* 117	0.894631E-04	0.178926E-03			
PSAR*	118100		0.330278E-02	0.434029E-04*	118
* 118	0.434029E-04	0.868059E-04			
PSAR*	119100		0.501557E-02	0.100092E-03*	119
* 119	0.100092E-03	0.200185E-03			
PSAR*	120100		0.330278E-02	0.434029E-04*	120
* 120	0.434029E-04	0.868059E-04			
PSAR*	121100		0.474179E-02	0.894631E-04*	121
* 121	0.894631E-04	0.178926E-03			
PSAR*	122100		0.377461E-02	0.566895E-04*	122
* 122	0.566895E-04	0.113379E-03			
PSAR*	123100		0.343494E-02	0.469460E-04*	123
* 123	0.469460E-04	0.938920E-04			
PSAR*	124100		0.343494E-02	0.469460E-04*	124
* 124	0.469460E-04	0.938920E-04			
PSAR*	125100		0.343494E-02	0.469460E-04*	125
* 125	0.469460E-04	0.938920E-04			
PSAR*	126100		0.343494E-02	0.469460E-04*	126
* 126	0.469460E-04	0.938920E-04			
PSAR*	127100		0.176541E-02	0.124008E-04*	127
* 127	0.124008E-04	0.248016E-04			
PSAR*	128100		0.176541E-02	0.124008E-04*	128
* 128	0.124008E-04	0.248016E-04			
PSAR*	129100		0.176541E-02	0.124008E-04*	129
* 129	0.124008E-04	0.248016E-04			
PSAR*	130100		0.176541E-02	0.124008E-04*	130
* 130	0.124008E-04	0.248016E-04			
PSAR*	181100		0.527517E-02	0.110722E-03*	181
* 181	0.110722E-03	0.221444E-03			
PSAR*	182100		0.412004E-02	0.675402E-04*	182
* 182	0.675402E-04	0.135080E-03			
PSAR*	183100		0.241739E-02	0.232515E-04*	183
* 183	0.232515E-04	0.465031E-04			
PSAR*	186100		0.412004E-02	0.675402E-04*	186
* 186	0.675402E-04	0.135080E-03			
PSAR*	187100		0.241739E-02	0.232515E-04*	187
* 187	0.232515E-04	0.465031E-04			
PSAR*	201100		0.377461E-02	0.566895E-04*	201
* 201	0.566895E-04	0.113379E-03			
PSAR*	202100		0.366236E-02	0.533679E-04*	202
* 202	0.533679E-04	0.106736E-03			
PSAR*	203100		0.151058E-02	0.907917E-05*	203
* 203	0.907917E-05	0.181583E-04			
PSAR*	204100		0.366236E-02	0.533679E-04*	204
* 204	0.533679E-04	0.106736E-03			
PSAR*	205100		0.151058E-02	0.907917E-05*	205

* 205	0.907917E-05	0.181583E-04				
PBAR*	207100		0.395462E-02	0.622256E-04*	307	
* 207	0.622256E-04	0.124451E-03				
PBAR*	232100		0.117957E-02	0.553610E-05*	232	
* 232	0.553610E-05	0.110722E-04				
PBAR*	239100		0.117957E-02	0.553610E-05*	239	
* 239	0.553610E-05	0.110722E-04				
PBAR*	62300		0.0060821	0.0015205	*	62
* 62	0.0015205	0.0030410	15.69			
PBAR*	116300		0.0060821	0.0015205	*	116
* 116	0.0015205	0.0030410	15.69			
PBAR*	184300		0.0060821	0.0015205	*	184
* 184	0.0015205	0.0030410	15.69			
PBAR*	185300		0.0060821	0.0015205	*	185
* 185	0.0015205	0.0030410	15.69			
PBAR*	131200		1.257E-3	3.142E-4		131
* 131	3.142E-4	6.285E-4	3.82			
PBAR*	132200		1.257E-3	3.142E-4*		132
* 132	3.142E-4	6.285E-4	3.82			
PBAR*	133200		1.257E-3	3.142E-4*		133
* 133	3.142E-4	6.285E-4	3.82			
PBAR*	134200		1.257E-3	3.142E-4*		134
* 134	3.142E-4	6.285E-4	3.82			
PBAR*	135200		1.257E-3	3.142E-4*		135
* 135	3.142E-4	6.285E-4	3.82			
PBAR*	136200		1.257E-3	3.142E-4*		136
* 136	3.142E-4	6.285E-4	3.82			
PBAR*	137200		1.257E-3	3.142E-4*		137
* 137	3.142E-4	6.285E-4	3.82			
PBAR*	138200		1.257E-3	3.142E-4*		138
* 138	3.142E-4	6.285E-4	3.82			
PBAR*	139200		1.257E-3	3.142E-4*		139
* 139	3.142E-4	6.285E-4	3.82			
PBAR*	140200		1.257E-3	3.142E-4*		140
* 140	3.142E-4	6.285E-4	3.82			
\$						
\$						
\$						
\$	MULTI-POINT CONSTRAINT FOR X-AXIS LOS ERROR (NODE 100 DOF 1)					
\$						
MPC*	100		100	1	-1.0*1000000	
*1000000		34	2	-0.01855287570	*1000001	
*1000001			34	3	-0.14285714286*1000002	
*1000002		35	2	-0.0185528757	*1000003	
*1000003			35	3	-0.14285714286*1000004	
*1000004		2830	3	0.28571428572	*1000005	
*1000005			30	3	0.0	*1000006
*1000006		27	2	0.08065681999	*1000007	
*1000007			27	3	-0.35489000795*1000008	
*1000008		29	2	0.08065681999	*1000009	
*1000009			29	3	-0.35489000795*1000010	
*1000010		3233	3	0.70978001590	*1000011	
*1000011			33	3	0.0	*1000012
*1000012		1002	4	-3.48423005566	*1000013	
*1000013			11	2	-0.06210394429*1000014	
*1000014		9	2	-0.06210394429		
\$						

```

$      MULTI-POINT CONSTRAINT FOR Y-AXIS LOS ERROR (NODE 100 DOF 2)
$
MPC#    100          100      2      -1.0*2000000
*2000000  34          1  -0.03710575139  *2000001
*2000001          34      2  -0.04638218924*2000002
*2000002          34      3  -0.2500000000  *2000003
*2000003          35      2  0.04638218924*2000004
*2000004          35      3  0.2500000000  *2000005
*2000005          27      1  0.16131363998*2000006
*2000006          27      2  -0.06049261499  *2000007
*2000007          27      3  -0.62105751391*2000008
*2000008          29      2  0.06049261499  *2000009
*2000009          29      3  0.62105751391*2000010
*2000010         1002      5  3.48423005566  *2000011
*2000011          11      1  -0.12420788859*2000012
*2000012          11      2  -0.07762993037  *2000013
*2000013          11      9  2  0.07762993037*2000014
*2000014
$
$      MULTI-POINT CONSTRAINT EQUATION FOR DEFOCUS (NODE 100 DOF 3)
$
MPC#    100          100      3      -1.0*3000000
*3000000  34          3  -0.01912393776  *3000001
*3000001          35      3  -0.01912393776*3000002
*3000002          2830      3  0.12749291836  *3000003
*3000003          30      3  0.0  *3000004
*3000004          27      3  0.77803217347  *3000005
*3000005          29      3  0.77803217347*3000006
*3000006          3233      3  -0.46681930408  *3000007
*3000007          1002      3  -0.17849008371*3000008
*3000008          9       3  0.50000000000  *3000009
*3000009          11      3  0.50000000000*3000010
*3000010          40      3  -2.000000000000
$
$      RIGID BODY SUPPORT
$
SUPORT,44,123456
ENDDATA

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